

# Formal Verification of Distributed Aircraft Controllers

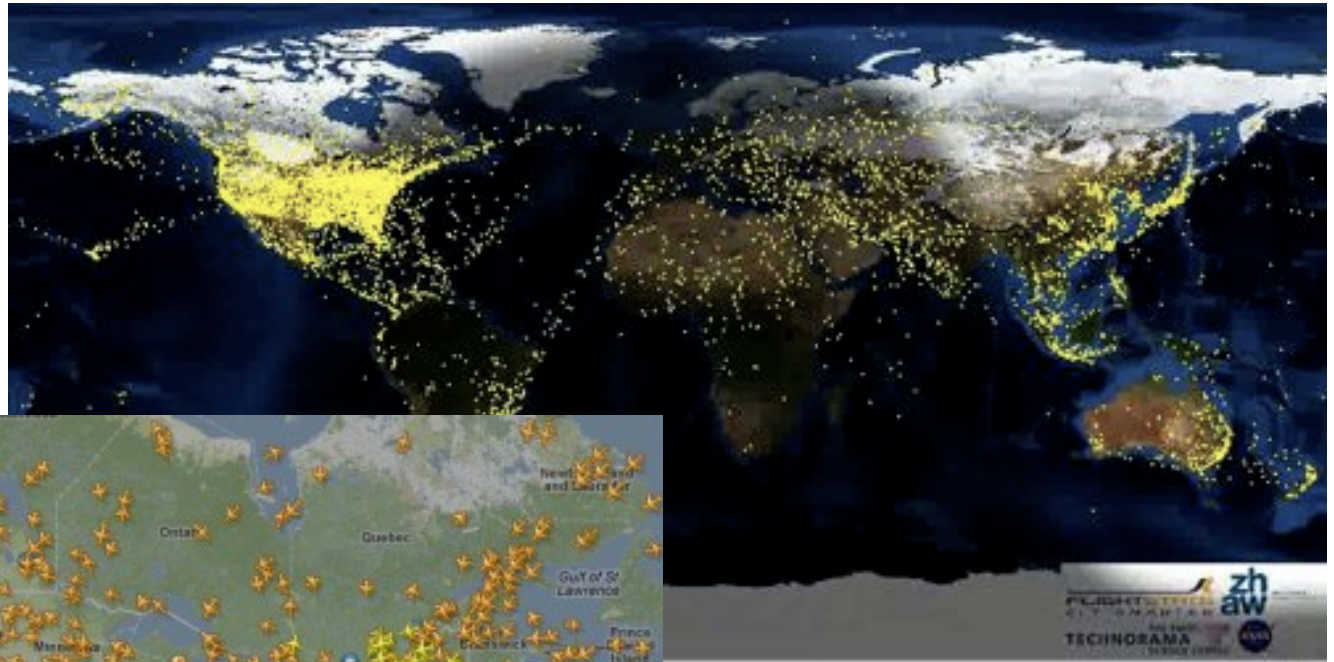
Sarah M. Loos, David Renshaw, and André Platzer

Computer Science Department

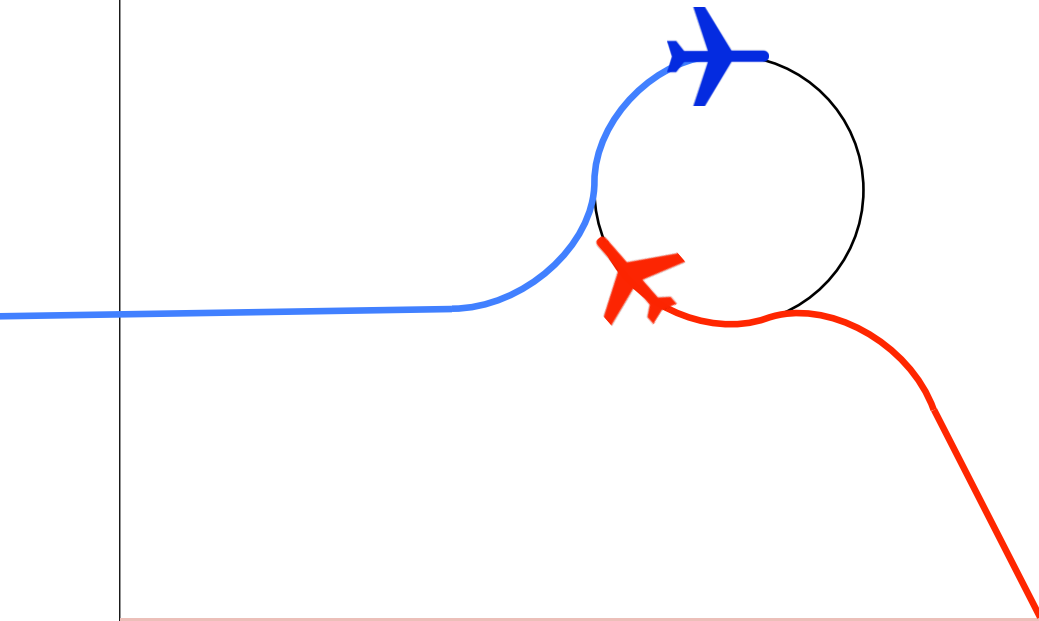
Carnegie Mellon University

April 10, 2013

# How Can We Prove Distributed Airspace?

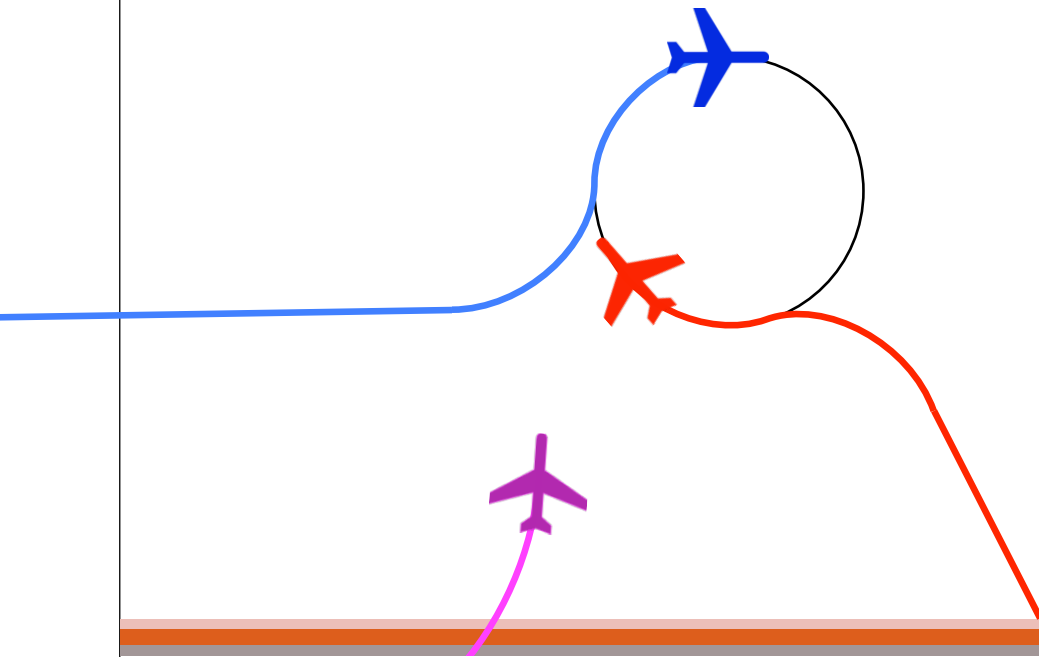


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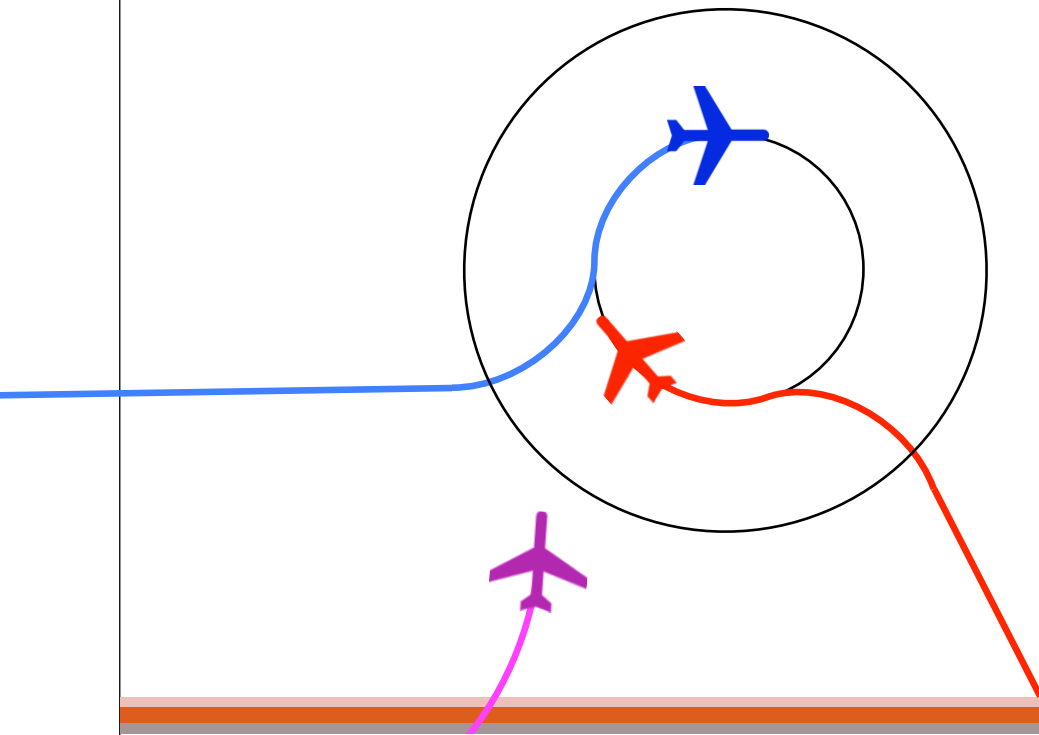
Sensor limits on aircraft are **local**.

# How Can We Prove Distributed Airspace?



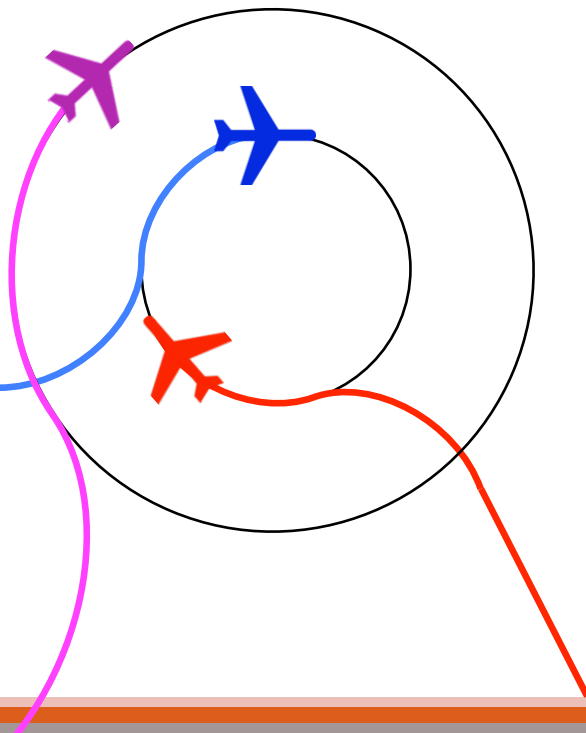
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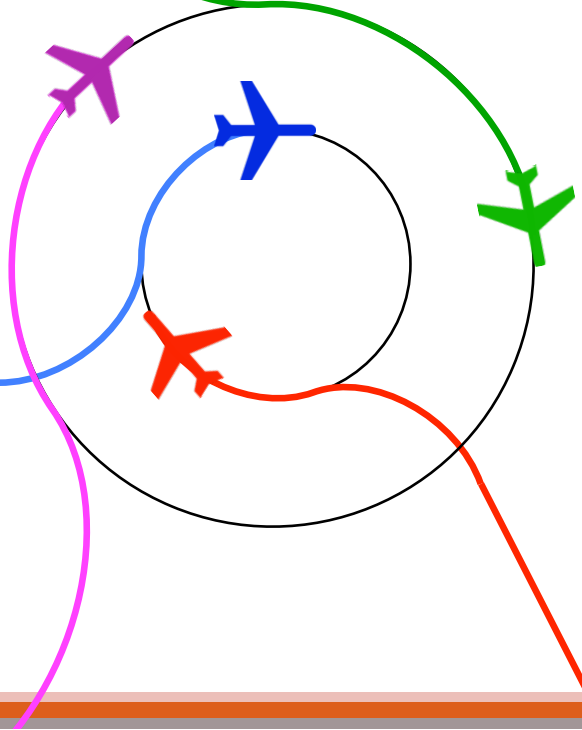
# How Can We Prove Distributed Airspace?



Sensor limits on aircraft are **local**.

Sometimes a maneuver may look safe **locally**...

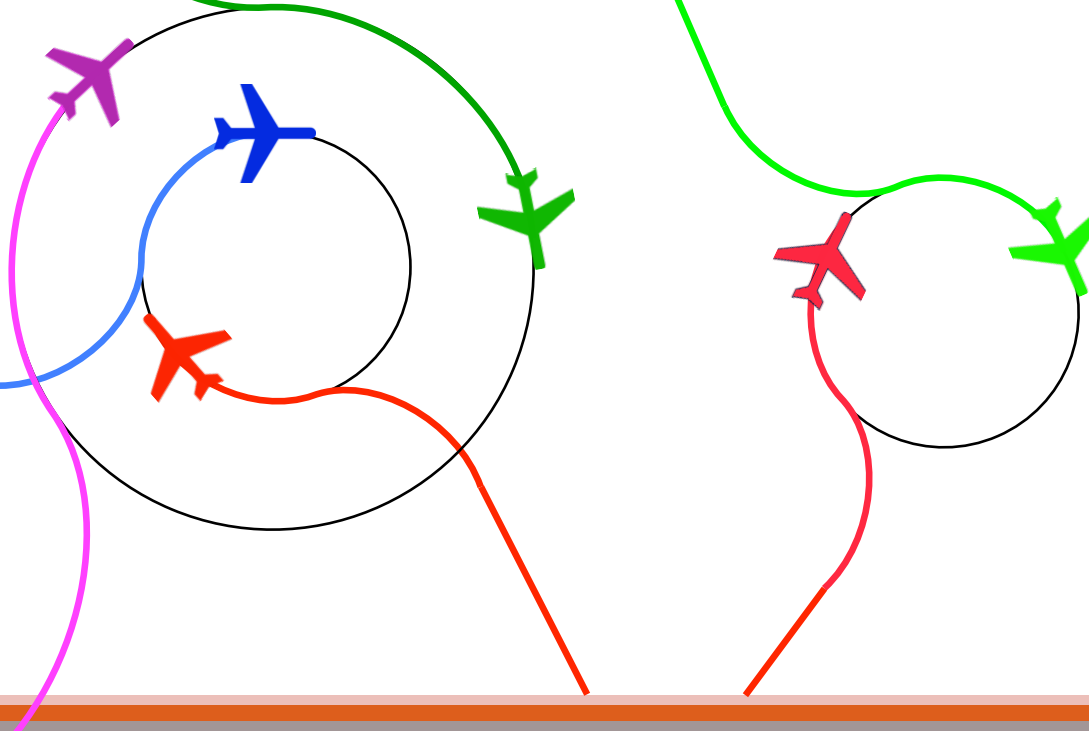
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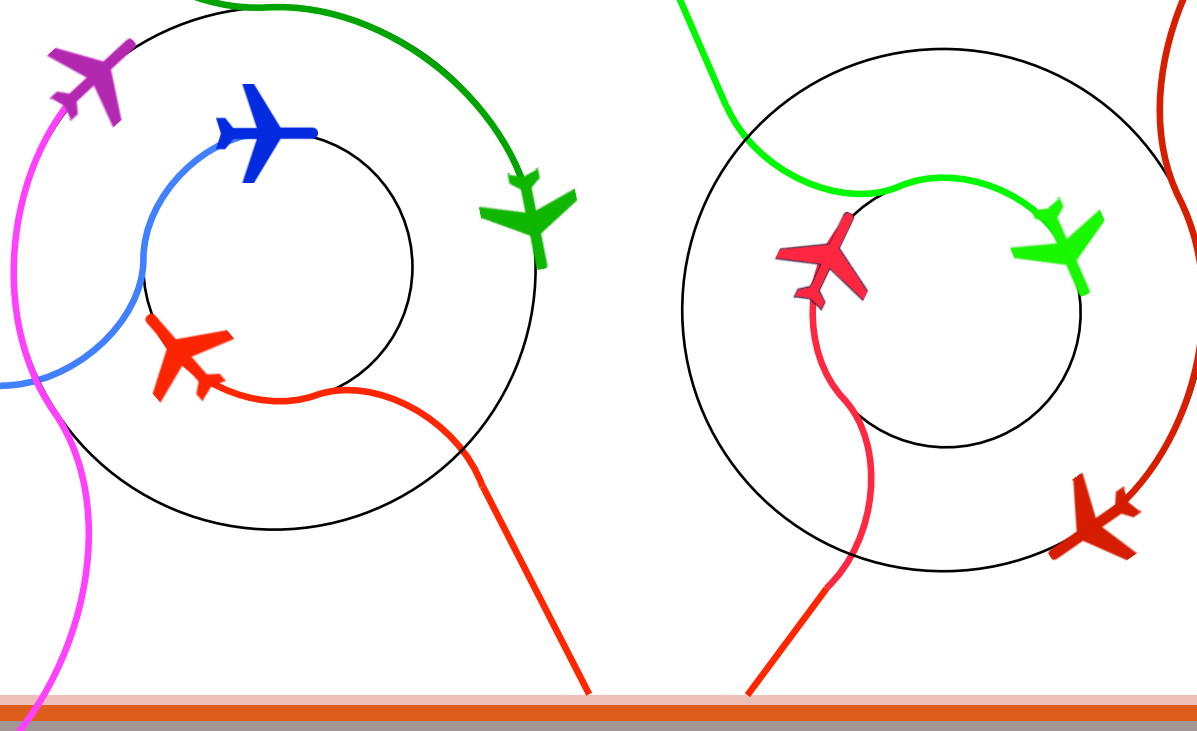
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Sometimes a maneuver may look safe **locally**...

But is a terrible idea when implemented **globally**.



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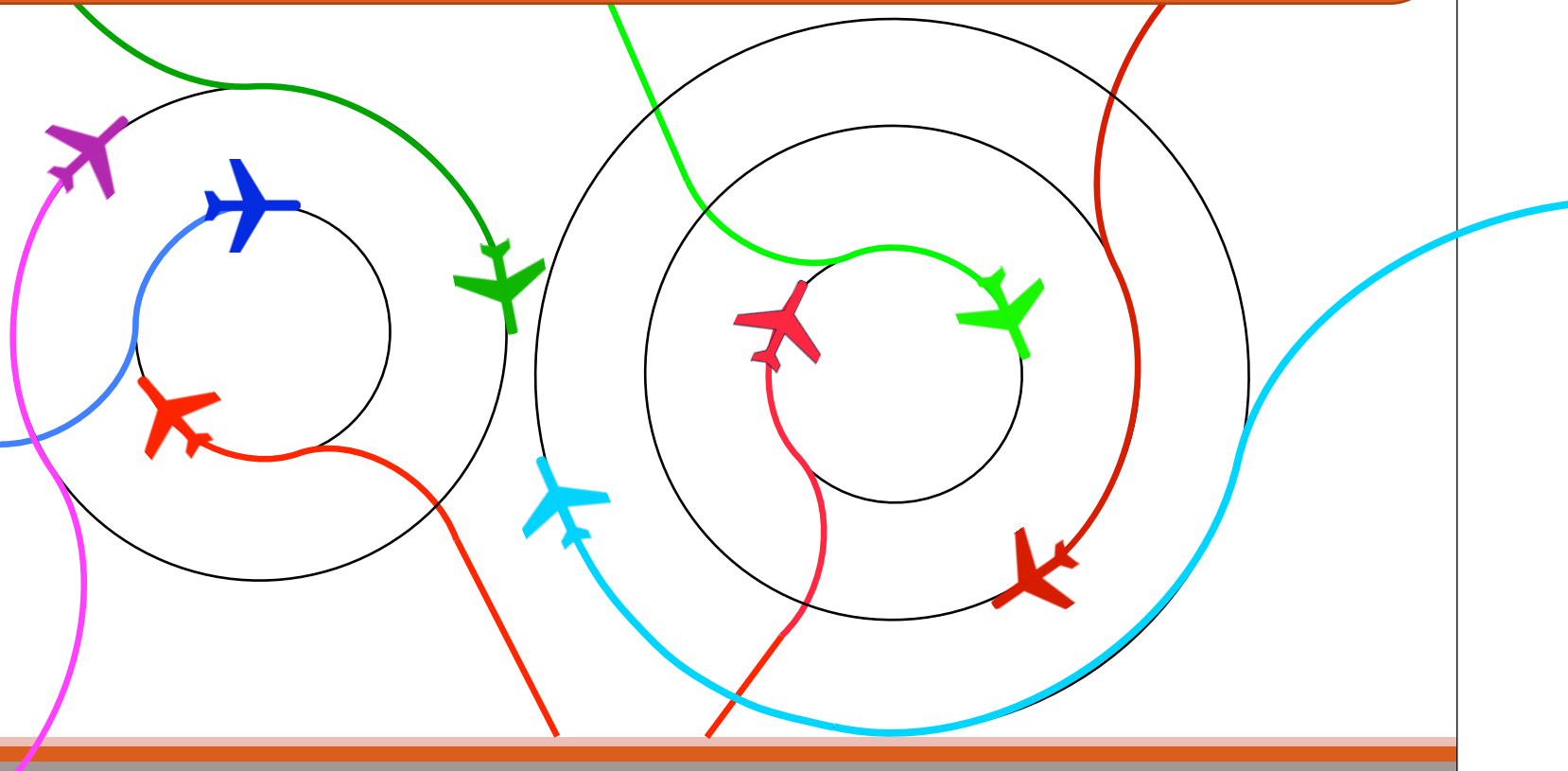


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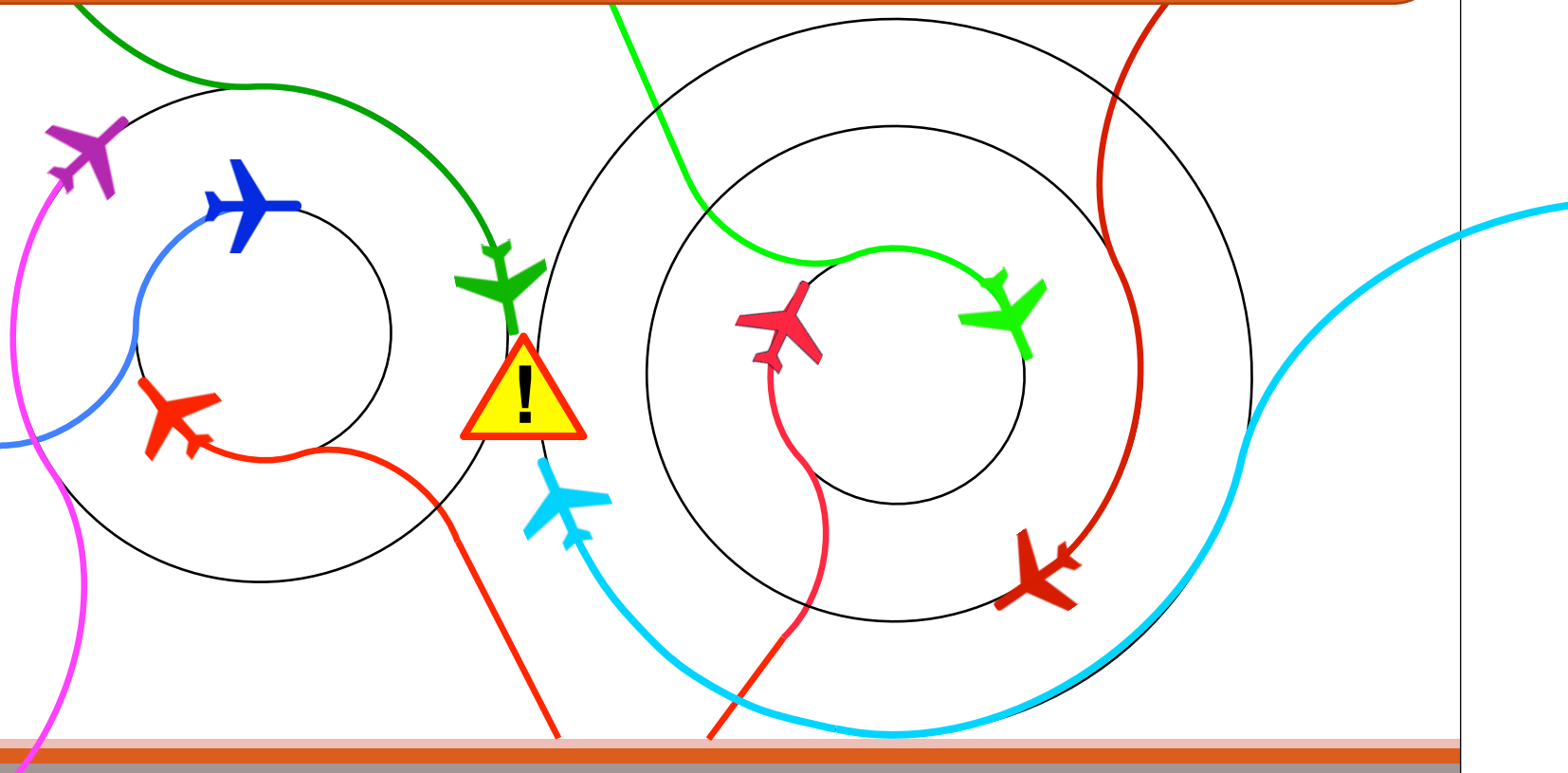


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# Assumptions and Requirements

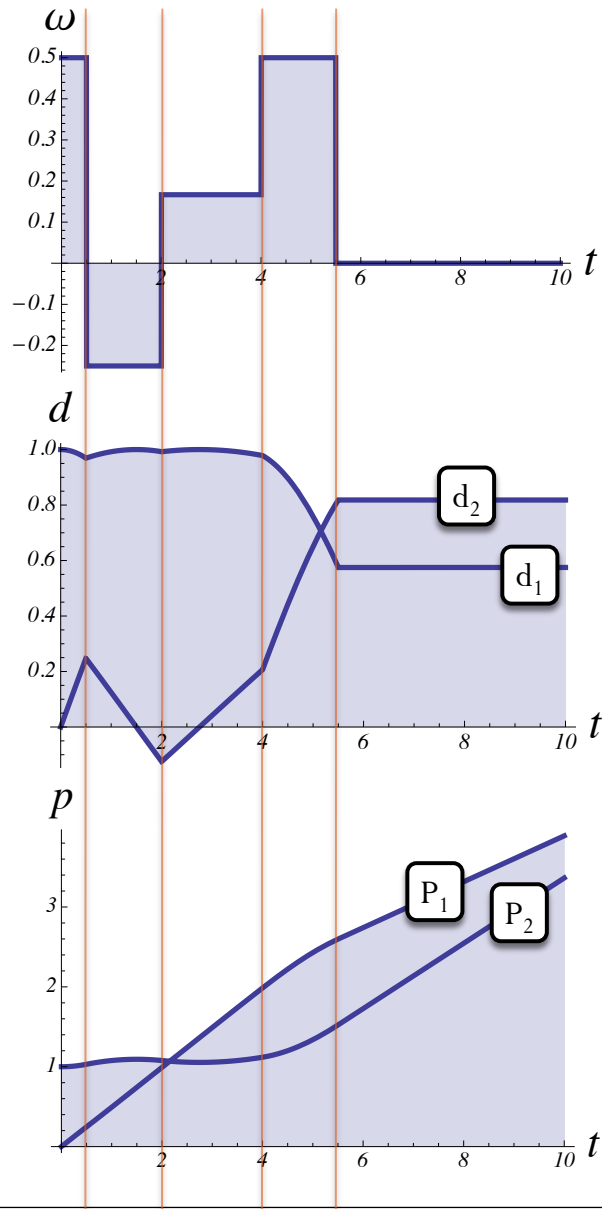
## Requirements

- **Safety:** At all times, the aircraft must be separated by distance greater than  $p$ .
- Aircraft trajectories must always be **flyable**.
- An **arbitrary number** of aircraft may enter the maneuver at any time.

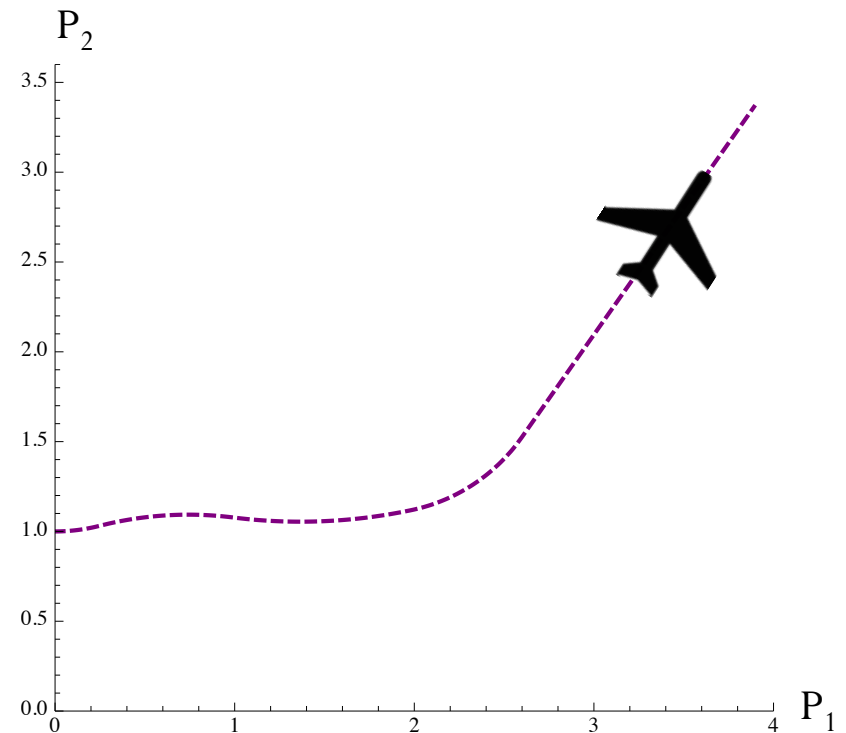
## Assumptions

- Aircraft maintain constant velocity.
- Sensors are accurate and have no delay.
- Collision avoidance maneuvers are executed on the 2D plane.

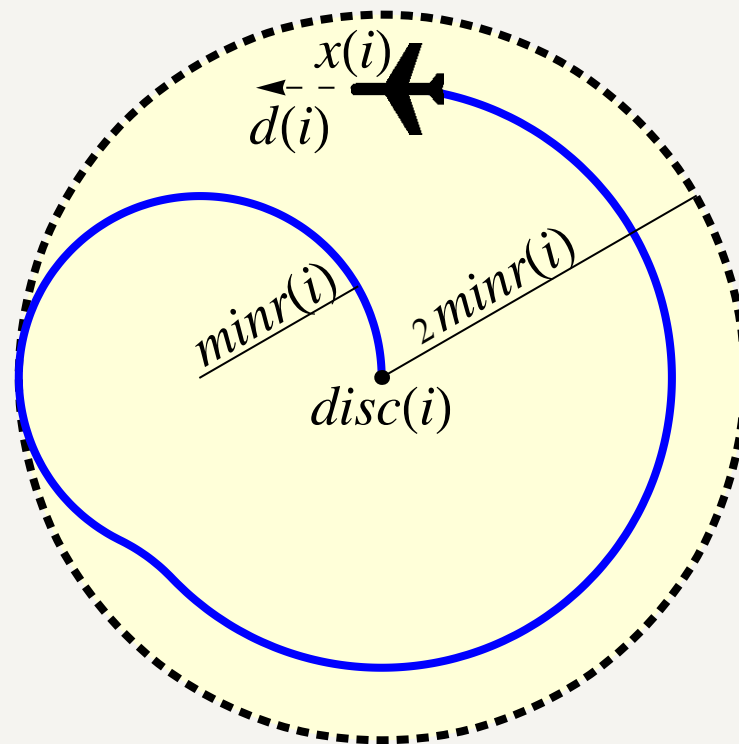
# Hybrid Dynamics



Aircraft are controlled by steering, through discrete changes in angular velocity  $\omega$ .



# Big Disc Control



- Leaves maneuverability to pilot discretion.
- Requires large buffer disc.
- Requires aircraft to return to the center of the disc before completing avoidance maneuver.

# Big Disc Control

$$\text{BigDisc} \equiv (\text{Control} \cup \text{Plant})^*$$

$$\text{Control} \equiv k := *_{\mathbb{A}}; (\text{CA} \cup \text{NotCA})$$

$$\text{CA} \equiv ?(ca(k) = 1); (\text{Steer} \cup \text{Exit})$$

$$\text{NotCA} \equiv ?(ca(k) = 0); (\text{Steer} \cup \text{Flip} \cup \text{Enter})$$

$$\text{Steer} \equiv \omega(k) := *_{\mathbb{R}}; ?(-\Omega(k) \leq \omega(k) \leq \Omega(k))$$

$$\text{Exit} \equiv ?(disc(k) = x(k)); ca(k) := 0$$

$$\text{Enter} \equiv \omega(k) := side(k) \cdot \Omega(k); ca(k) := 1$$

$$\text{Flip} \equiv side(k) := -side(k)$$

$$\text{Plant} \equiv \forall i : \mathbb{A} \left( x(i)' = v(i) \cdot d(i), d(i)' = \omega(i) \cdot d(i)^\perp, \right. \\ \left. disc(i)' = (1 - ca(i)) \cdot v(i) \cdot d(i) \ \& \ \text{EvDom} \right)$$

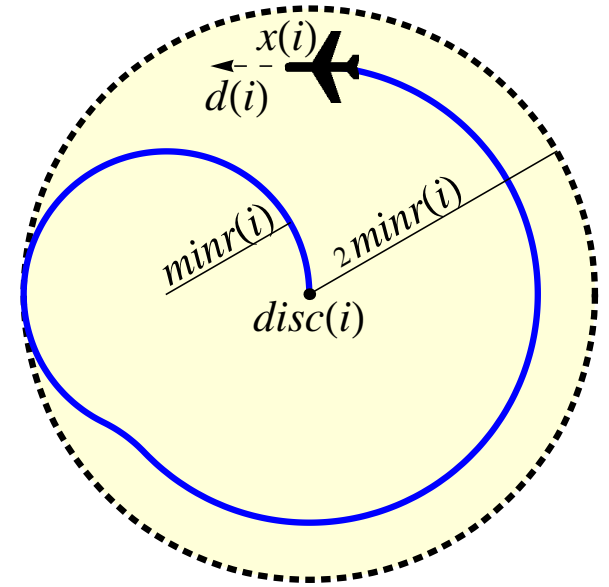
$$\text{EvDom} \equiv \forall j : \mathbb{A}$$

$$((j \neq i \wedge (ca(i) = 0 \vee ca(j) = 0)) \rightarrow \text{Sep}(i, j)$$

$$\wedge \|disc(i) - (x(i) + minr(i) \cdot side(i) \cdot d(i)^\perp)\|$$

$$\leq minr(i))$$

$$\text{Sep}(i, j) \equiv \|disc(i) - disc(j)\| \geq 2minr(i) + 2minr(j) + p$$



# Big Disc Control

BigDisc  $\equiv$  (Control  $\cup$  Plant)\*

Control  $\equiv k := *_A; (CA \cup \text{NotCA})$

CA  $\equiv ?(ca(k) = 1); (\text{Steer} \cup \text{Exit})$

NotCA  $\equiv ?(ca(k) = 0); (\text{Steer} \cup \text{Flip} \cup \text{Enter})$

Steer  $\equiv \omega(k) := *_R; ?(-\Omega(k) \leq \omega(k) \leq \Omega(k))$

Exit  $\equiv ?(disc(k) = x(k)); ca(k) := 0$

Enter  $\equiv \omega(k) := side(k) \cdot \Omega(k); ca(k) := 1$

Flip  $\equiv side(k) := -side(k)$

Plant  $\equiv \forall i : \mathbb{A} (x(i)' = v(i) \cdot d(i), d(i)' = \omega(i) \cdot d(i)^\perp,$

**✓ Verified in KeYmaeraD**

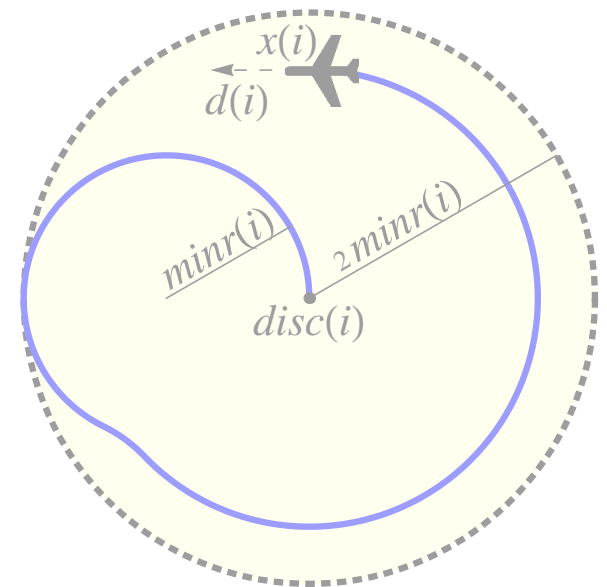
EvDom  $\equiv \forall j : \mathbb{A}$

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$\wedge \|disc(i) - (x(i) + minr(i) \cdot side(i) \cdot d(i)^\perp)\|$

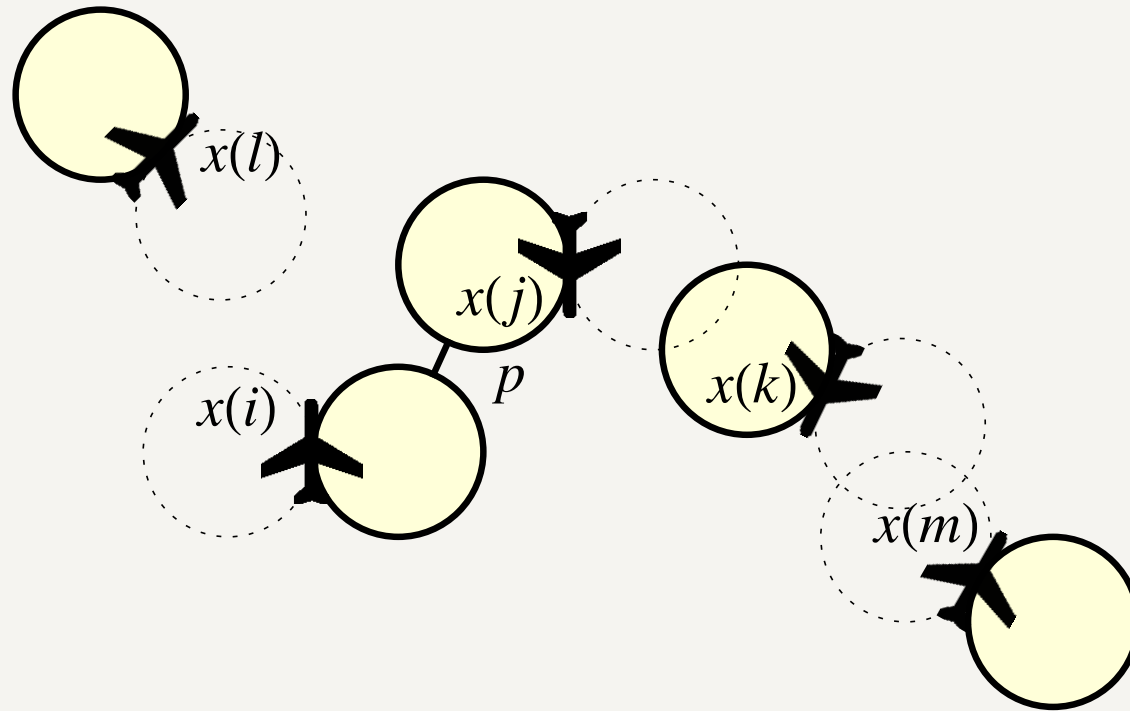
$\leq minr(i))$

Sep( $i, j$ )  $\equiv \|disc(i) - disc(j)\| \geq 2minr(i) + 2minr(j) + p$





# Small Discs Control



- Deterministic control makes it well suited for UAVs.
- Smaller discs allow aircraft to fly closer together.
- Aircraft may exit maneuver as soon as it is safe to do so.

# Small Discs Control

SmallDiscs  $\equiv$  (Control  $\cup$  Plant)\*

Control  $\equiv$   $k := *_{\mathbb{A}}$ ; (CA  $\cup$  NotCA)

CA  $\equiv$   $?(ca(k) = 1)$ ; (Exit  $\cup$  Skip)

NotCA  $\equiv$   $?(ca(k) = 0)$ ; (Steer  $\cup$  Flip  $\cup$  Enter)

Skip  $\equiv$   $?true$

Steer  $\equiv$   $\omega(k) := *_{\mathbb{R}}$ ;  $?(-\Omega(k) \leq \omega(k) \leq \Omega(k))$

Exit  $\equiv$   $ca(k) := 0$

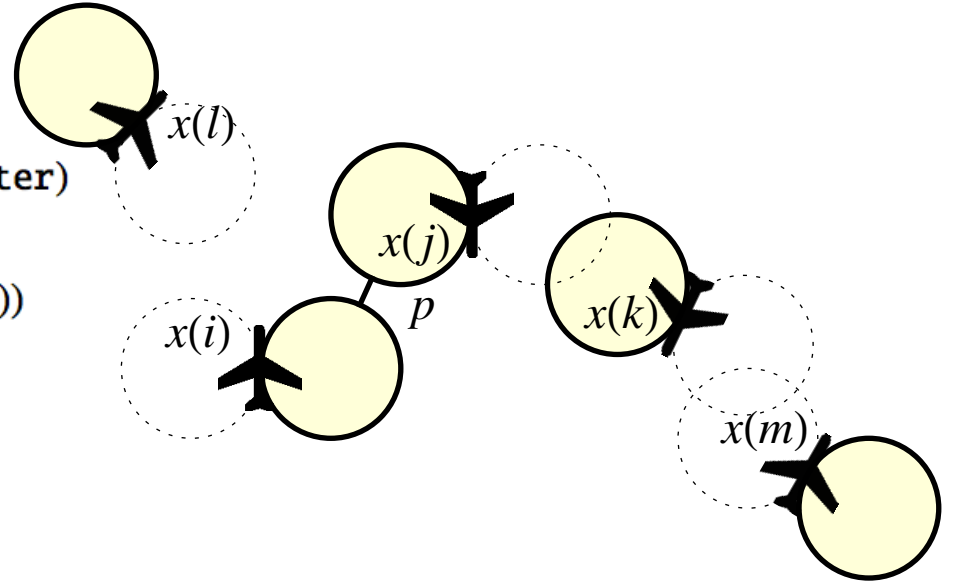
Enter  $\equiv$   $(\omega(k) := side(k) \cdot \Omega(k)); ca(k) := 1$

Flip  $\equiv$   $?(\forall j : \mathbb{A} (j \neq k \rightarrow FlipSep(j, k)))$ ;  
 $side(k) := -side(k)$

FlipSep( $i, j$ )  $\equiv$   $\| (x(i) + minr(i) \cdot side(i) \cdot d(i)^\perp$   
 $- (x(j) - minr(j) \cdot side(j) \cdot d(j)^\perp) \|$   
 $\geq minr(i) + minr(j) + p$

Plant  $\equiv$   $\forall i : \mathbb{A} (x(i)' = v(i) \cdot d(i), d(i)' = \omega(i)d(i)^\perp$   
 $\& \forall j : \mathbb{A} ((j \neq i \wedge (ca(i) = 0 \vee ca(j) = 0))$   
 $\rightarrow Sep(i, j))$

Sep( $i, j$ )  $\equiv$   $\| (x(i) + minr(i) \cdot side(i) \cdot d(i)^\perp$   
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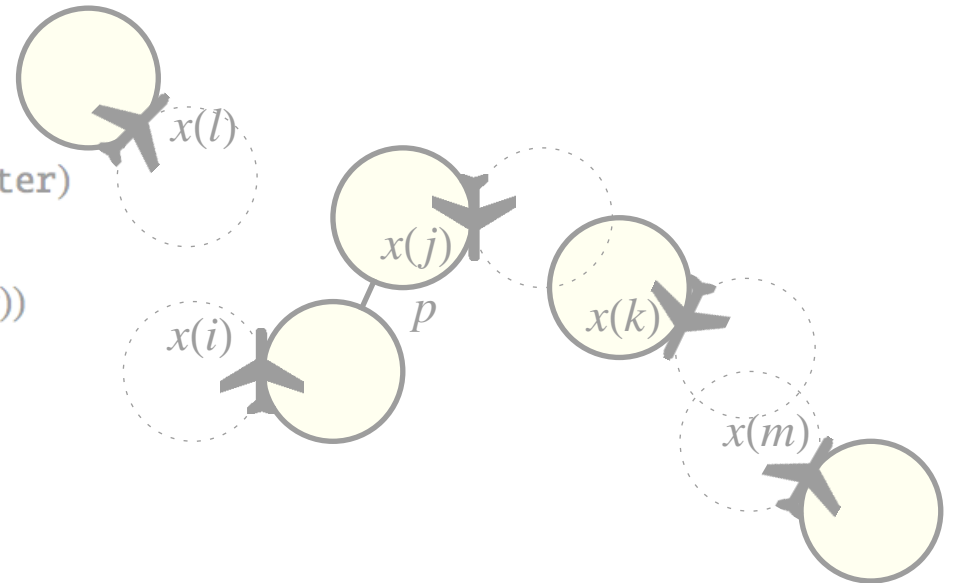
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$- (x(j) + minr(j) \cdot side(j) \cdot d(j)^\perp) \|$

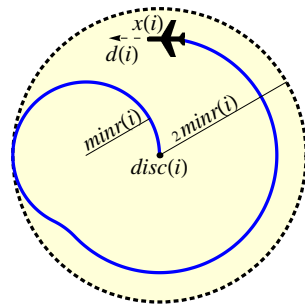
$\geq minr(i) + minr(j) + p$



# Conclusions

## Challenges

- Infinite, continuous, and evolving state space,  $\mathbb{R}^\infty$
- Continuous dynamics
- Discrete control decisions
- Distributed dynamics
- Arbitrary number of aircraft
- Emergent behaviors



## Solutions

- Quantifiers for distributed dynamics
- Compositionality – using small problems to solve the big ones
- Hierarchical and modular proofs
- Non-linear flight paths allow flyable maneuvers

